

Short timescale imaging polarimetry of geostationary satellite Thor-6: the nature of micro-glints

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Abstract

Large constellations of orbiting communication satellites will become an important source of noise for present and future astronomical observatories. Mitigation measures rely on high quality predictive models of the position and expected brightness of these objects. Optical linear imaging polarimetry holds promise as a quantitative tool to improve our understanding of the physics of reflection of sunlight off satellite components and through which models of expected brightness can be improved. We present the first simultaneous short-timescale linear polarimetry and optical photometry observations of a geostationary satellite, using the new MOPTOP imaging polarimeter on the 2m Liverpool Telescope. Our target, telecommunication satellite Thor-6, shows prominent short timescale glint-like features in the lightcurve, some as short as seconds. Our polarimetric observations overlap with several of these micro-glints, and have the cadence required to resolve them. We find that the polarisation lightcurve is remarkably smooth, the short time scale glints are not seen to produce strong polarimetric features in our observation. We show how short timescale polarimetry can further constrain the properties of the components responsible for these micro-glints.

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1. Introduction

The characterisation of the reflection of sunlight by orbiting artificial satellites has become an increasingly important and urgent field of research in recent years, not least because of the rapid build-up of large mega-constellations of communication satellites. The reflected light of these objects is bright

enough to noticeably impact sensitive astronomical observations (e.g. McDowell, 2020; Hainaut & Williams, 2020; Rawls et al., 2021) at a large range of wavelengths, not just on the ground but also from low-earth orbit. Predictive models of both the ephemerides and the expected brightness of satellites are therefore of crucial importance to predict, evaluate and potentially mitigate their impact on sensitive astronomical observations (e.g. Hainaut & Williams, 2020). Most of the current efforts have focussed on obtaining (multi-colour) broadband flux

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lightcurves (e.g. Horiuchi et al., 2020; Tregloan-Reed et al., 2021; Mróz et al., 2022), and basic models have been created to evaluate expected brightness as a function of sun-observer-satellite angle (e.g. Hainaut & Williams, 2020; Mallama, 2020; Cole, 2021; Bassa et al., 2022; Lawler et al., 2022).

Many satellites show glint features in their lightcurve, during which their brightness dramatically increases during a short period of time. Glints form through specular (or near-specular) reflection from relatively flat reflective parts of the satellite (e.g. solar panels) at a specific range of sun-satellite-observer angles. Satellite glints can be mistaken for astronomical sources (e.g. Schaefer et al., 1987) and form an undesirable foreground in short timescale transient searches (e.g. Corbett et al., 2020; Karpov et al., 2019).

Many of the satellites that are of greatest concern to astronomical observatories show brightnesses close to the detector saturation point of sensitive astronomical telescopes, and glinting may therefore form an additional risk factor (e.g. Hainaut & Williams, 2020). The timescales of glint features are determined by the rate of change of geometry, e.g. in rotating bodies glints are very short. The shape of the reflecting features also imprints on the glint duration. Some satellites show a variety of glint timescales and amplitudes (e.g. Hall & Kervin, 2013; Chote et al., 2019). As shown by Vrba et al. (2009), an ideal flat reflector on a geostationary orbit produces a glint that lasts around ~ 2 minutes. Many observed glints last significantly longer than this (with timescales of around an hour), and show lower peak amplitudes than in the ideal reflector case. This indicates that the reflecting components giving rise to the glint, e.g. a solar panel, is not an ideal flat but for example consist of multiple flat pieces that are somewhat tilted with respect to one another (e.g. Vrba et al. 2009). Some geostationary satellites show glint-like features in their lightcurves with durations much shorter than traditional glints. In the following we will refer to those as micro-glints, for which we adopt a working definition of glint-like brightenings with durations below 2 minutes in geostationary orbit. Glints (and micro-glints) are not just a nuisance, but can also form a valuable tool to inform models of satellite reflections (e.g. Hall & Kervin, 2013). Polarimetry directly diagnoses the orientation as well as the material properties of the reflecting surfaces, it can therefore solve many of the existing degeneracies in glint models, and provide the necessary physical parameters needed for quantitative analytical modelling of reflection of satellites, both in glint phases and outside of glints.

Reflection of light off a surface induces linear polarisation. The resultant wavelength-dependent polarisation degree and polarisation angle are strong functions of the angle of reflection and the physical properties of the reflecting material. The latter are captured in the complex index of refraction n_c , which is defined in terms of the refractive index n and the extinction coefficient k as $n_c = n - i * k$; the linear polarisation induced by specular reflection will be maximal at the Brewster angle of the reflecting material. Satellites consist of several reflecting surfaces, with different relative orientations and with different refractive indices n_c . The main reflecting surfaces are the solar panels, the side(s) of the spacecraft bus that faces the ob-

server (which may be covered in multi-layer insulation, MLI), and the antenna dishes. As a satellite orbits the Earth, the angle of the sunlight reflecting of different elements rapidly changes: we should therefore see changing polarisation degree and angle as a function of time. When the reflection angle gets close to the Brewster angle of the material of a reflecting component, we may expect a strong change in the total observed polarisation. The polarisation properties of spacecraft materials have been studied numerically and in the lab (e.g. Pasqual & Cahoy, 2017; Beamer et al., 2018; Peltoniemi et al., 2021). In principle, the problem can be reversed, and the satellite's orientation and physical parameters of reflection can be empirically determined from well-sampled multi-colour polarimetric lightcurves (polarisation degree and angle), assuming some basic shape properties and geometry (aided by lightcurve analysis, e.g. Seo et al., 2013) as priors, by fitting a Mueller matrix chain (describing the optical action of each reflecting element) directly onto the total observed wavelength-dependent polarisation as a function of angle. This is a method frequently used in calibration and design of optical telescopes and instruments, where we can fit for the orientation and indices of refraction of reflecting surfaces as free parameters in the components of the Mueller matrix chain made up of all optical components (see e.g. Wiersema et al., 2018, for an example). To do this successfully for satellites requires multi-wavelength, high cadence, high accuracy (low systematic errors) polarimetry over a substantial range of solar phase angles (the Sun-object-observer angle). Such datasets are not yet publicly available. However, single wavelength, lower cadence polarimetry datasets are an important first step, to identify the main satellite components responsible for the observed polarisation (e.g. Speicher et al., 2015; Beamer et al., 2018; Kosaka et al., 2020), to provide an inventory of empirical polarisation behaviour for a variety of satellite platforms (e.g. Speicher et al., 2015) and to postulate a sensible range of priors for more quantitative fitting methods.

Glints are particularly helpful lightcurve features (e.g. Vrba et al., 2009), as these are expected to show substantial amounts of optical linear polarisation (Speicher et al., 2015; Zimmerman et al., 2020). They are bright, which reduces statistical errors of polarisation measurements. While some polarimetric data exists of glints (e.g. Zimmerman et al., 2020), short duration events like micro-glints are not well studied polarimetrically to date. Speicher et al. (2015) have shown indications of optical polarimetric signals associated with micro-glints, but their study was limited to relatively long lasting micro-glints (several minutes) studied at relatively poor temporal resolution, with generally only one or two polarimetric datapoints covering the lightcurve feature. To use micro-glints as a quantitative tool, we need polarimetry at timescales of seconds, with small polarimetric uncertainties ($\sigma_P \lesssim 0.2\%$).

The data discussed in this paper were taken as part of a pilot programme to use a new imaging polarimeter (MOPTOP, the Multi-colour OPTimised Optical Polarimeter; Jermak et al., 2016, 2018; Shrestha et al., 2020) on the robotic Liverpool Telescope (Steele et al., 2004) to study changes in orientation of satellites through their polarisation signatures, particularly the docking of the MEV-2 vehicle with the geostationary Intelsat

130 10-02 satellite. During that programme, we observed another
 131 geostationary satellite, Thor-6, as a calibration observation (i.e.
 132 a secondary calibrator): that observation is the topic of this pa-
 133 per. This object was selected because of its close proximity on
 134 the sky to the MEV-2 + Intelsat 10-02 pair, and its well mon-
 135 itored lightcurves (Chote et al. in prep.). Thor-6 is also inter-
 136 esting in its own right: this satellite shows bright and frequent
 137 micro-glints in its optical lightcurves, and therefore enables a
 138 first search for polarimetric signals of micro-glints at timescales
 139 of a few seconds in reflected optical light of geostationary satel-
 140 lites. While geostationary satellites generally do not pose a risk
 141 to astronomical observations (in contrast to satellite constella-
 142 tions at lower orbits), they are a useful testbed for the type of
 143 observational studies required to better understand the reflec-
 144 tion properties of satellites that do pose a risk but are more chal-
 145 lenging to study, e.g. because of their rapid movement on the
 146 sky (beyond the maximum non-sidereal tracking speed of many
 147 older $\gtrsim 2$ m class telescopes).

148 In this paper we show our acquisition, analysis and cali-
 149 bration of the MOPTOP polarimetry of Thor-6, as an exam-
 150 ple of the capabilities of MOPTOP for short timescale optical
 151 polarimetry of moving objects, and compare the data to simul-
 152 taneous lightcurves. We show how our data, and future data
 153 covering a larger range of time, can be used to place constraints
 154 on (or measure directly) the nature of the structural components
 155 of the satellite causing micro-glints.

156 2. Thor-6 and polarimetry of geostationary satellites

157 Thor-6 (also known as Intelsat 1W) is a currently active
 158 geostationary telecommunication satellite, primarily providing
 159 television broadcasting services. It is owned by Telenor Satel-
 160 lite Broadcasting AS, and built by Thales Alenia Space. It was
 161 launched on 29 October 2009 by an Ariane 5ECA launch ve-
 162 hicle, from Kourou, French Guyana. Thor-6 uses the Thales
 163 Alenia Spacebus-4000B2 platform. Its shape is broadly of the
 164 “box-wing” type: a box-shaped bus, several large dish antennas
 165 and two long rectangular solar panels extending from the north
 166 and south faces of the bus, spanning a few tens of meters.

167 Geostationary satellites have been studied using optical po-
 168 larimetry before. These observations were generally performed
 169 at relatively low cadence. Speicher et al. (2015) observed a
 170 small sample of geostationary objects using a polarimeter on a
 171 small telescope, finding a relatively large diversity in polarisa-
 172 tion lightcurves, likely reflecting a diversity in satellite shape
 173 and geometry. The authors used an instrument that recorded
 174 two channels, a horizontally and a vertically polarised compo-
 175 nent. Based on changes of the relative strength of the horizon-
 176 tal and vertical components as a function of viewing geometry,
 177 some inferences can be made to which satellite component is
 178 contributing most polarised light. Zimmerman et al. (2020) ob-
 179 served a small sample of geostationary satellites with a small
 180 telescope, using quasi-simultaneous polarimetric and low res-
 181 olution spectroscopic observations, finding evidence for an in-
 182 crease in linear polarisation during times that a glint was visi-
 183 ble in the lightcurve. For a satellite in low-Earth orbit we ex-
 184 pect similar polarimetric behaviours (after correcting for orbit

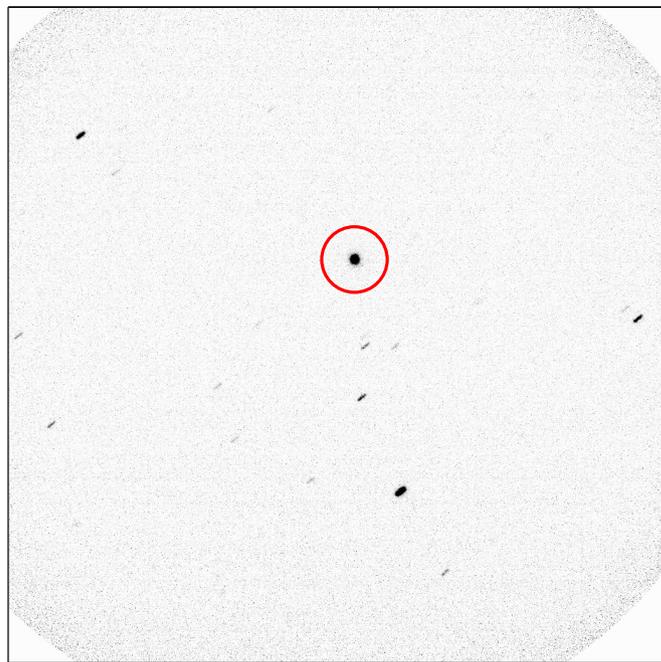


Fig. 1. A full, representative, single MOPTOP image from our observation, with Thor-6 circled. This is image 1_e_20210504_14_26_4, i.e. an image of cam1, with run number 14, rotation number 26 and waveplate position 4. The integration time for this image is 0.4 seconds (the fixed value for FAST mode observations). Stars can be seen as streaks in the image. Because of the short integration time of individual images, these streaks are relatively short.

185 orientation differences: geostationary satellites are located in
 186 a fairly narrow equatorial belt, whereas low earth orbit satel-
 187 lites cover a wide range of inclinations), with the key difference
 188 that they traverse the range of solar phase angles over a much
 189 shorter timespan, compressing the relevant timescales, which
 190 makes obtaining diagnostic data more challenging.

191 Kosaka et al. (2020) observed geostationary satellite
 192 Express-AM5 using a much larger telescope, the 2m Nayuta
 193 telescope, and a polarimeter, for ~ 5 hours at a cadence of
 194 90 seconds, forming one of the highest quality and highest cadence
 195 polarimetric datasets of a geostationary satellite to date.
 196 In their data, they see a minimum in the optical polarisation
 197 ($P_{\text{lin}} \sim 1\%$) around the time of the minimum phase angle, with
 198 rapidly increasing linear polarisation after the minimum phase
 199 angle (with values increasing up to $P_{\text{lin}} \sim 14\%$ in the phase an-
 200 gle interval covered by their observations). The measurements
 201 from Kosaka et al. (2020) provide full Stokes Q, U, I (see Sec-
 202 tion 4) and are calibrated onto the absolute polarisation degree
 203 and angle values system (IAU, 1973, where polarisation angle
 204 towards North is 0° and East is 90°). However, at 90 second
 205 cadence, rotation of the satellite with respect to the observer can
 206 be significant and cause artefacts in the data; observations at
 207 higher time resolution are needed to avoid these.

208 3. Observations

209 The observations reported in this paper consist of a 800 sec-
 210 ond high-cadence imaging polarimetry observation taken with

the MOPTOP instrument on the Liverpool Telescope, and a high-cadence optical lightcurve taken simultaneous with the polarimetry, from the same geographical location, using the University of Warwick test telescope.

3.1. MOPTOP observations

The polarimetric observation of Thor-6 in this paper was performed robotically by the 2m Liverpool Telescope (LT; Steele et al. 2004), located on the island of La Palma, Spain; under proposal number DL21A02 (PI Wiersema). We used the Multi-colour OPTimised Optical Polarimeter (MOPTOP) imaging polarimeter (Jermak et al., 2016, 2018; Shrestha et al., 2020). This dual beam polarimeter, optimised for time-domain astrophysics, uses a continuously rotating half-wave plate and a wiregrid polarising beamsplitter; two scientific CMOS cameras (Andor Zyla sCMOS cameras) record the images of the two orthogonally polarised beams simultaneously, hereafter we refer to these two cameras as *cam1* and *cam2*. The detector readouts are synchronised to the waveplate rotation. Sixteen images are recorded by each camera for every full (360 degree) waveplate rotation; for details and design motivation see Jermak et al. (2016, 2018) and Shrestha et al. (2020). A total of 32 images are therefore recorded for each full waveplate rotation, at mean waveplate angles of $0^\circ, 22.5^\circ, \dots, 337.5^\circ$. MOPTOP can be used with two fixed wave plate rotation speeds, the SLOW and FAST mode. In the former, the rotation period of the wave plate is 80 seconds, in the latter 8 seconds. This translates to a frame exposure time of 4.0 s in SLOW mode, and 0.4 s in FAST mode (the remaining time is used for read-out). For the observation discussed in this paper we used the FAST rotator observing mode, which is best suited to bright sources and provides good time resolution. The MOPTOP observations of Thor-6 used a R filter (MOP-R), covering the wavelength range $\sim 580 - 695$ nm.

The observation of Thor-6 was prepared and executed as follows: in the afternoon before the night of observation, we retrieved the most recent Two-Line Elements (TLEs) for the target from the Celestrak website (<https://www.celestrak.com/NORAD/elements/>). We then generated an ephemeris table for the geographical location and altitude of the Liverpool Telescope (LT) using the JPL Horizons On-Line Ephemeris System, with a time resolution of 1 minute. Within the LT phase2 tool, the target was uploaded as an ephemeris table (a so-called *ephemeris target*), and observations were defined using the FIXED observing mode, i.e. a fixed time was defined at which the observations were to be started, within a user-configurable tolerance (the so-called *slack*, which we set at 10 minutes for this observation). The resulting predefined observation was entered into the LT queue, with no constraints placed on the seeing and sky brightness; observations were selected, scheduled and executed robotically. When executing a given ephemeris target observation, the telescope will interpolate between the coordinates given in the ephemeris file for acquisition and tracking. Note that LT can not auto-guide on moving objects. The observation was taken with the Cassegrain mount angle rotation set to zero degrees. The airmass for this observation was 1.275; the first exposure was started at 03:19:08.511 UT on

5 May 2021, and we observed for a total on-target time of 800 seconds (100 wave plate rotations; this timespan is currently the limit for a single FAST mode observation). The weather conditions were good and the seeing at the start of the polarimetric observation was $\sim 1''$. The solar declination at the start of the observation was $+16.207$ degrees.

3.2. Photometric observations

We obtained a large number of high-cadence optical light curves between February and May 2021 as part of an observation campaign studying the rendezvous, proximity operations, and docking of MEV-2 with Intelsat 10-02 (see George et al., 2021). Observations were made using the University of Warwick's test telescope, also located on La Palma, which was configured for these observations using a Takahashi Epsilon 180ED wide-field astrograph with an Andor Marana sCMOS detector. This combination provided a $2.6^\circ \times 2.6^\circ$ field of view with a pixel scale of $4.5''/\text{pixel}$. Simultaneous full-night light curves were obtained for Intelsat 10-02, MEV-2, Thor-5, Thor-6, and Thor-7, which are located close together on the sky, all within the field of view of this telescope. We used cadences between 1 s and 5.5 s (the shortest exposures were necessary to avoid saturation during the main glint features around local midnight, where brightness could peak as high as 5th magnitude). The full observation campaign and data reduction procedures will be discussed in a future publication (Chote et al. in prep); in this paper we will use the lightcurve coinciding with the MOPTOP polarimetric observations, which is shown in Figure 2. The photometry is calibrated by integrating over the streaks of an ensemble of suitable calibration stars (selected to avoid blending with other star streaks, of suitable brightness, and non-variable) and matching the instrumental magnitude against Gaia to obtain a zero point in Gaia G and a colour term that is evaluated at $(G_{BP} - G_{RP})_{\odot} = 0.82$, the Gaia colour of the Sun (Casagrande & Vandenberg, 2018). Typically around 400 calibration stars are used per image. The light curve (Figure 2) is plotted as a function of the Solar equatorial phase angle, which is defined as the longitudinal component of the angle between the satellite and the anti-solar point (Payne et al., 2007).

4. MOPTOP data reduction and analysis

The MOPTOP data reduction procedure is described by Shrestha et al. (2020) and at the MOPTOP website. The raw frames undergo bias and dark subtraction, and are corrected using a flatfield constructed from a stack of flatfield images at all 16 waveplate positions (i.e. the flatfield is the same for the images at all 16 waveplate positions, see Shrestha et al., 2020). This method works well for dual beam polarimeters under certain conditions (for a discussion see e.g. Patat & Romaniello, 2006).

Our analysis procedure of the reduced data (i.e. measuring fluxes and computing polarisation) differs slightly from the methods set out in Shrestha et al. (2020), and we detail our approach in the following. First, the reduced data are sorted by date and epoch and some basic properties of the data are retrieved from the file headers. The centroid of the target is then

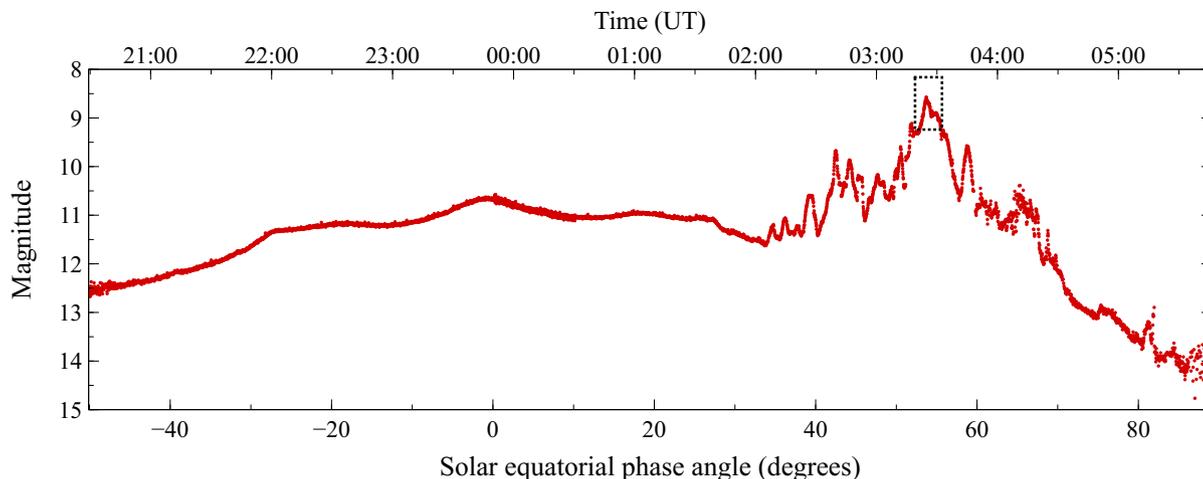


Fig. 2. Optical lightcurve of Thor-6 in the night starting 4 May 2021 (see Section 3.2). Both the solar equatorial phase angle (defined as the longitudinal component of the angle between the satellite and the anti-solar point; Payne et al. 2007) and the time of observation (UT) are shown on the horizontal axes. The box marks the timespan of the MOPTOP observations, shown in detail in Figure 4. Magnitudes are in the Vega system, calibrated onto Gaia G band values for field stars (Chote et al. in prep.).

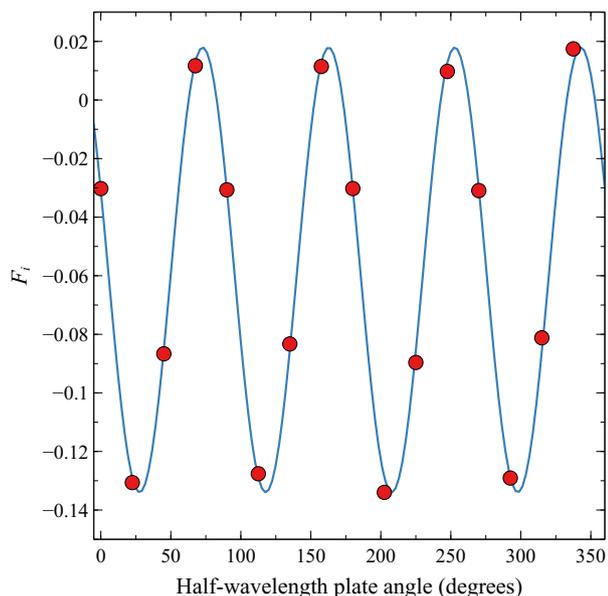


Fig. 3. The red symbols show the normalised flux differences F_i of one of our observations (a single full waveplate rotation) of a polarised standard star with MOPTOP in FAST mode, using the MOP-R filter. The blue solid line is the sum of the $n = 0$ and $n = 4$ Fourier components (Section 4).

measured using the IRAF *starfind* and *imcentroid* tasks. As geostationary objects maintain a broadly constant altitude and azimuth, stars in the field move rapidly over the detector, forming streaks (Fig 1), which in rare cases may influence the centroiding when they happen to pass very close to the target. We use fairly strict sigma-clipping values in the centroiding procedure to eliminate this effect; our target is very bright. We measure the fluxes of the target in the *cam1* and *cam2* images, using aperture photometry in IRAF, using the *apphot* package. The aperture radius is chosen as 2 times the average FWHM (full width at half maximum) of a Gaussian fit to the object point spread function (the target is unresolved), and is kept fixed for all exposures: the seeing was stable during the MOPTOP observation duration to within 0.1 arcsecond. Aperture radii are the same for *cam1* and *cam2*. An annulus shaped region was used to determine the local sky background level. Hereafter we use the notation $f_{\text{cam1},i}$ and $f_{\text{cam2},i}$ for the target flux in camera 1 and camera 2 at the i -th waveplate angle. We compute normalised flux differences $F_i = (f_{\text{cam1},i} - f_{\text{cam2},i}) / (f_{\text{cam1},i} + f_{\text{cam2},i})$ for each exposure set at each angle ϕ_i of the half-wave plate.

First, we analyse a set of polarised standard stars (three observations of HD 155197, one of Hiltner 960 and one of VI Cyg 12), all observed in FAST mode and using the MOP-R filter (these stars span the magnitude range 10.6-9.4 mag; standard sidereal tracking is used for these observations). We measure their fluxes in the same way as for Thor-6, and compute their normalised flux differences F_i . We then perform a simple Fourier analysis on the standard star F_i values to verify the modulation behaviour of MOPTOP, following Patat & Taubenberger (2011), using the expression (Fendt et al., 1996; Patat & Taubenberger, 2011):

$$F_i = a_0 + \sum_{n=1}^{N/2} [a_n \cos(n(2\pi i/N)) + b_n \sin(n(2\pi i/N))],$$

where a_n, b_n are the Fourier coefficients, N the number of wave-

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plate angles, and i the i -th angle as above. As explained in Patat & Taubenberger (2011), an ideal dual beam polarimeter of the design of MOPTOP would have all its Fourier power in the $n = 4$ component, and all other components would be zero. We fit the F_i data of the polarised standard stars using this Fourier prescription, using the *symfit* package (Roelfs & Kroon, 2020) in Python. As expected, we find that the only statistically significant terms (found with $\gtrsim 5\sigma$ significance) are the $n = 0$ (i.e. a_0) and the $n = 4$ terms. Pleochroism ($n = 2$ component) is not significantly detected. Figure 3 shows an example MOPTOP MOP-R band FAST mode dataset of polarised standard star Hiltner 960 (observed on 8 May 2021), where a model consisting only of the $n = 0$ and $n = 4$ terms is shown to provide an excellent description of the data. We describe the polarisation state of incoming light through the Stokes vector $\vec{S} = (I, Q, U, V)$; note that some authors prefer the equivalent notation $\vec{S} = (S_0, S_1, S_2, S_3)$ for the Stokes vector components. In the following we will not consider the Stokes V (or S_3) component: in reflection scenarios as we consider here, optical circular polarisation is mainly caused by cross-talk, i.e. circular polarisation is induced when the reflected light was somewhat linearly polarised before reflection, so there is cross-talk between the Q, U and V Stokes parameters. This can for example take place in scenarios where light gets reflected twice, or in reflection from complex layered materials. We therefore generally expect low values of circular polarisation, and in the following we focus on the linear polarisation.

Given the result of the Fourier analysis above, we use a simple prescription for calculating the Stokes parameters as:

$$q = Q/I = \frac{2}{N} \sum_{i=0}^{N-1} F_i \cos\left(\frac{i\pi}{2}\right) \quad (1)$$

$$u = U/I = \frac{2}{N} \sum_{i=0}^{N-1} F_i \sin\left(\frac{i\pi}{2}\right) \quad (2)$$

for each set of 4 waveplate angles, i.e. we compute four independent values for q, u for each full waveplate rotation. In other words, waveplate angles $0^\circ, 22.5^\circ, 45^\circ$ and 67.5° give q_1, u_1 ; $90^\circ, 112.5^\circ, 135^\circ$ and 157.5° give q_2, u_2 ; etc. We use these independent measurements as our individual datapoints, giving a time resolution of 2 seconds. The errors on the Stokes parameters are calculated through standard error propagation. The values are corrected for instrumental polarisation using the values for the MOP-R band listed on the MOPTOP website ($q_{\text{inst}} = +0.0091, u_{\text{inst}} = -0.0302$ for MOP-R), which we verified using a MOPTOP dataset of an unpolarised standard star taken close in time to the Thor-6 observation. We compute the linear polarisation P_{lin} and the polarisation angle θ via

$$P_{\text{lin}} = \sqrt{q^2 + u^2} \quad (3)$$

$$\theta = \frac{1}{2} \arctan\left(\frac{q}{u}\right), \quad (4)$$

where the quadrant-preserving arctan is used. In the conversion from q, u to P_{lin}, θ we expect to encounter the effects of polarisation bias (Serkowski, 1958; Wardle & Kronberg, 1974;

Simmons & Stewart, 1985). This bias arises from the fact that q and u can be positive or negative, with their errors generally a Normal distribution. In contrast, P_{lin} is positive definite (equation 3), and has a Ricean probability distribution. In the presence of noise on q and u , we can therefore over-estimate P_{lin} in situations with low signal to noise, this is often referred to as polarisation bias. There are a large number of studies offering various correction techniques to take this into account. We use the modified asymptotic (MAS) estimator, as defined in Plaszczyński et al. (2014) to correct for polarisation bias, but find that in all observations of Thor-6 the polarisation bias plays no significant role (this is not surprising: the flux signal to noise f/σ_f is very high, as is the polarisation signal to noise P/σ_P). The resulting polarisation values are corrected for instrumental de-polarisation, for which we use the multiplicative value tabulated on the MOPTOP website for the MOP-R band, as derived from observations of polarised standard stars, which we verified using the observations of polarised standard stars mentioned above. The final calibration step consists of placing the polarisation angle in the correct absolute frame, for which we follow the prescription from the MOPTOP website: $\theta_{\text{true}} = \theta_{\text{inst}} + \theta_{\text{rotskypa}} + c$, where θ_{inst} is the instrumental polarisation angle found above, θ_{rotskypa} is the instrument rotation angle as tabulated in the *rotskypa* header keyword, and c is a constant offset. We use the polarised standard star observations to compute the average offset between instrumental polarisation angle (corrected for their θ_{rotskypa} values) and their values from the literature: we used the values tabulated in Schmidt et al. (1992) for HD 155197, Hiltner 960 and VI Cyg 12. Note that Hiltner 960 and VI Cyg 12 may show some signs of variability (Blinov et al., 2021). We find a 1.0° systematic error on the absolute angle values for our set of standard star observations. This calibration should place the polarisation on the IAU definition of polarisation angle (IAU, 1973).

The final polarisation lightcurve, of both linear polarisation degree and angle, with the individual 2 second datapoints and a 4-point binned average (8 seconds), is shown in Figure 4.

5. Discussion

5.1. Polarimetry of moving objects with LT+MOPTOP

The polarisation lightcurve of Thor-6 (Figure 4) shows that polarimetry at short timescales of moving objects with magnitudes typically seen for active geostationary satellites is indeed feasible with the MOPTOP instrument on LT. The field of view of MOPTOP ($\sim 7' \times 7'$) is large enough to reliably place a moving object in the field of view, if the object has a relatively recent TLE. Our observation in this paper, and those of MEV-2 (Wiersema et al. in prep) show that in most cases, moving targets can be placed close to the optical axis by the robotically operated LT, where the instrumental polarisation is well calibrated (Shrestha et al., 2020). Observations of much faster moving objects, such as the Starlink satellites in low-earth orbit, are more challenging for LT, as their angular velocity over the sky exceeds the current limits of the telescope (J. Marchant, priv. comm.). Trained imaging polarimetry (where

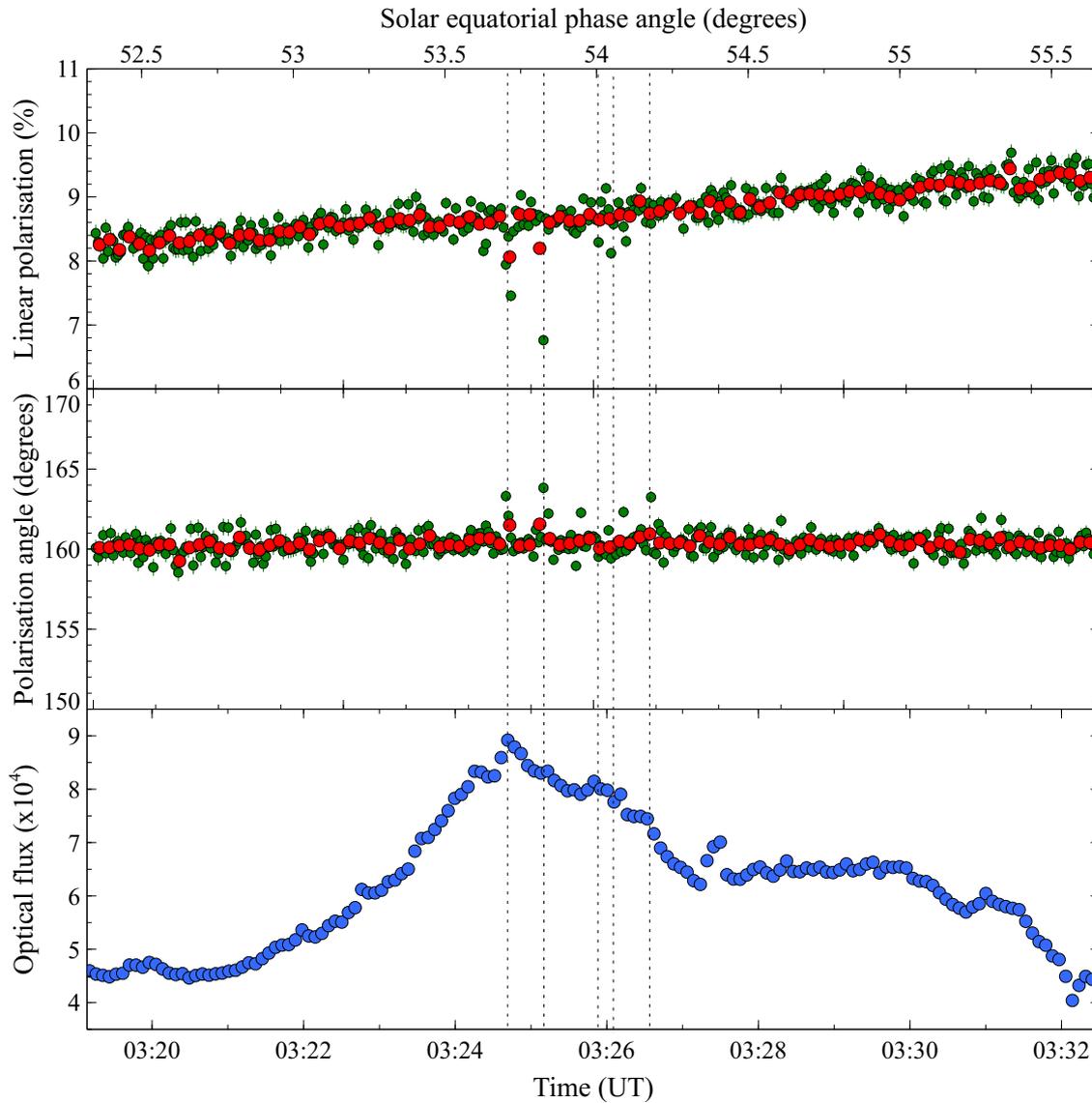


Fig. 4. The polarisation lightcurve from the 800s MOPTOP observation described in this paper, with the linear polarisation degree (P_{lin}) in the top panel, and the calibrated polarisation angle in the middle panel. Green points are the individual, independent measurements, red points are 4-point binned values. The bottom panel shows the optical lightcurve during the same time interval, here shown in linear flux values (in analog-to-digital units, ADU) as measured by the Warwick test telescope (see section 3.2) rather than magnitudes (Fig. 2), to allow a more intuitive comparison with the linear polarimetry. Several lightcurve features are clearly visible at short and longer timescales, with a range of amplitudes. The thin dashed vertical lines indicate the position of some of the outlier polarisation datapoints. The plot symbol size is larger than the formal errorbars for the lightcurve data and the binned polarimetry.

the satellite creates trails in the images) may be possible in some cases, but at the expense of significantly increased systematic errors. This tracking speed limit is not a limitation for some other observatories and commonly used mounts, and a short-timescale imaging polarimetry campaign is important to better inform efforts to mitigate against the impact of mega-constellations on astronomical observations. Another important property is the magnitude of the satellite: brighter objects allow the use of shorter exposures and waveplate rotation timescales for the polarimeter for a given σ_P requirement. In many cases, observations at larger phase angles hold important diagnostic power, but satellites are generally fainter than (Figure 2); telescopes of $\sim 2\text{m}$ class play an important role to provide accurate high-cadence polarimetry in those cases. Thor-6 was magnitude $\sim 9.4 - 8.6$ during the interval covered by MOPTOP. This gave good statistical errors (of order 0.1%, i.e. similar to the MOPTOP systematic errors, Shrestha et al., 2020). Even at the peaks of the observed glint signatures, the peak of the target point spread function was relatively far from image saturation or non-linearity limits, indicating that somewhat brighter glints can still be safely observed by MOPTOP in FAST mode.

The scatter of the datapoints around the general trend in Figure 4 is somewhat larger than one might expect based on the formal statistical errors of the individual datapoints, indicating some non-optimal effects play a role. One of those is the drift of the target over the detector: in an ideal polarimeter, the target would always occupy the same pixels, so that the beam-swapping that takes place by using four waveplate angles (equations 1 and 2) minimizes the effects of imperfect flatfielding, and so that hot pixels and other defects can be more efficiently corrected for. In our dataset, we see some drift of the target over the image during the observation. Figure 5 shows the centroid of Thor-6 move in a fairly monotonic fashion in X and Y pixel coordinates, mostly along the North-South direction, moving Northwards. The total position change in the 800 second observation is approximately 10.8 arcseconds. Comparison with the calibrated astrometry from the Warwick test telescope (Section 3.2) shows that this drift is primarily caused by errors in the TLE orbit prediction, rather than faults in the telescope tracking (LT can not auto-guide on moving objects). At the timescale of the individual sets of 4 waveplate angles that make up one q, u measurement (2 seconds) the drift is negligible, and the target point spread function is well described by a Gaussian profile throughout. Another possible reason for additional scatter (and potentially a fraction of the outlier datapoints) is the rapid passage of field stars through the source aperture or the sky annulus region (see Figure 1). Observatories with poorer resolution (large effective pixel scales) suffer this effect more than ones with better resolution. The spatial resolution of MOPTOP is excellent (0.42" per pixel, seeing of 1"), so this will only affect a very small number of datapoints. Visual inspection of the exposures confirms that this is indeed not the cause of the outlier datapoints (see Figure 6 for an example), the aperture and annulus radii are relatively small and the satellite did not cross through dense star fields in our observing time. Finally, the sCMOS detectors used on MOPTOP show "popcorn" noise (see Shrestha et al., 2020): random telegraph noise appearing as

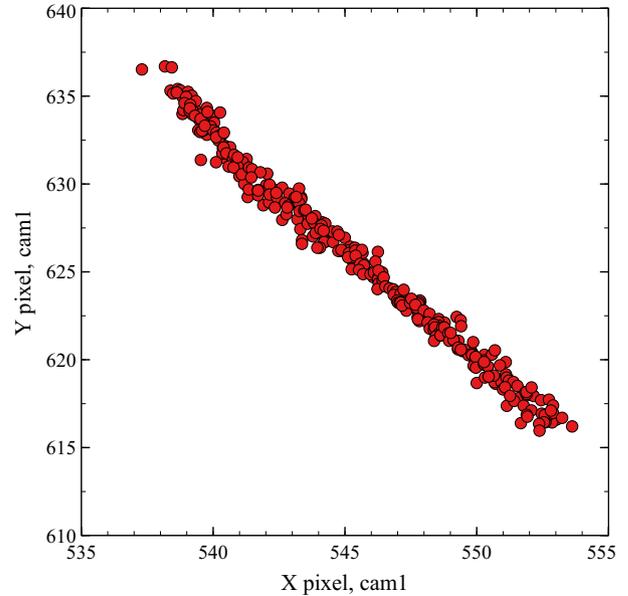


Fig. 5. The centroid pixel position of Thor-6 in the *cam1* data. The position gradually and monotonically drifts from top left to bottom right in this diagram. This is almost entirely along the North-South direction (specifically, moving northwards). The pixel scale of MOPTOP is 0.42"/pixel, the total drift is $\sim 10.8''$.

hot pixels at random locations in each frame. These may alter flux measurements somewhat when appearing by chance in the source aperture. We use four waveplate angles for each single q, u pair measurement, this beam-swapping reduces the influence of such single pixel noise events in single images. Combining more than four angles to make one q, u measurement further reduces this influence, but this comes at the cost of temporal resolution. In Figure 4 we therefore show the single (2 sec) datapoints and a binned version averaging four datapoints to one point.

5.2. Micro-glints and Thor-6

The MOPTOP polarimetric lightcurve of Thor-6 (Figure 4) shows a relatively smooth trend, with a broadly linear increase in polarisation degree P_{lin} from $\sim 8.2\%$ to $\sim 9.3\%$ in the 800 s covered by our observation. During this interval, the polarisation angle θ stays broadly constant. On top of this long-timescale trend only some low-amplitude wiggles may be present, limited to low polarimetric amplitudes ($\lesssim 0.3\%$) and short timescales (tens of seconds at most) on top of the general trend. The slowly increasing polarisation and the slow evolution of the polarisation angle in the MOPTOP interval is reminiscent of the polarisation curve of the geostationary satellite Express-AM5, presented by Kosaka et al. (2020) (their Figure 2), which had very similar P_{lin} and slowly varying polarisation angle at the phase angle of the MOPTOP observations of Thor-6 ($\sim 52.32 - 55.66$ degrees, Fig 4). The polarimetric lightcurve in Kosaka et al. (2020) is sampled at much lower temporal resolution (90 seconds vs the MOPTOP 2 seconds), but is of longer duration. The authors attribute the majority of the behaviour of P_{lin} in their dataset of Express-AM5 to reflection of the so-

lar panels of the satellite, with possible smaller contributions from the bus and/or the antenna dishes. This is based on the behaviour of the polarisation angle with time and the value of the linear polarisation as a function of reflection angle compared to the values found in the lab by Beamer et al. (2018). It is important to point out that the Express-AM5 satellite has a different platform from Thor-6, though shares many of the main features, e.g. a box-like bus, large antennas and large extended wing-like solar panels. We consider it likely that the longer timescale trend in the MOPTOP polarisation data of Thor-6 is similarly caused by reflection of the solar panels, using the same arguments as made by Kosaka et al. (2020).

Thor-6 is an interesting target because of the presence of short duration, bright, glint-like flares in the high cadence lightcurves (Chote et al. in prep) on top of the smoother diffuse reflection. Additional multi-filter observations (Chote et al. in prep) showed that there were no significant colour changes associated with these features. This means we can reliably compare features in our wide-band lightcurves with the R band polarimetry. Figure 2 shows the lightcurve at the night of the MOPTOP observations (i.e. this lightcurve was taken simultaneous with the polarimetry, with a telescope at nearly the same geographical location). Up until ~ 02 UT the lightcurve is smooth, showing only broad features, with a peak near zero phase angle, which is commonly seen in lightcurves of geostationary satellites. After ~ 02 UT a phase of rapid lightcurve variability starts, with many short duration, overlapping, glint-like peaks, some only barely resolved at the cadence of our lightcurve observations. Two broader features, at around $\sim 45^\circ$ and $\sim 54^\circ$ solar phase angle seem present, with many short peaks superposed on them. These short glints, which we call micro-glints here, appear to have a wide distribution of amplitude, duration and shape, with some lasting considerably shorter than a minute.

The lower panel of Figure 2 shows the small portion of the lightcurve which covers the time interval of the MOPTOP polarimetry. In Figure 4 we show that same lightcurve in instrumental flux units (analogue to digital units, ADU) rather than magnitudes, together with the polarisation lightcurve, to allow easy comparison; the MOPTOP data cover several micro-glints with a range of amplitudes and timescales. It is important to note here that the LT and the Warwick telescope are on the same mountain peak, separated by just ~ 250 meters. This is smaller than the expected size of the glint patch striking the Earth for an ideal flat reflector (Vrba et al., 2009). We can therefore directly compare the lightcurve and the polarisation.

Firstly we note that some outlier datapoints (or regions of larger scatter) are visible in the unbinned polarisation lightcurves. A subset of these may be attributable to some instrumental noise effects (Section 5.1), but it is clear that they take place near the peaks of the highest amplitude micro-glints, which may indicate a causal relation. There is also some indication that some more gradual changes/ripples (spanning $\sim 10 - 30$ sec) in the polarisation degree (with amplitudes $\sim 0.3\%$) occur at the time of some of the micro-glints, e.g. near the peak of the prominent micro-glint at $\sim 53.7^\circ$ phase angle (~ 337 sec in Figure 4). However, several other micro-

glints seen in the flux lightcurve seem not to have produced a detectable polarimetric feature, for example the one at $\sim 54.4^\circ$ phase angle.

The duration and amplitude of glint features from an ideal flat reflector is given by the crossing time of the sun spot size at the observer (e.g. Vrba et al., 2009). In many cases, glints in (not rapidly rotating) geostationary satellites appear to take much longer than this, which is generally attributed to a non-ideal reflector, e.g. a panel made up of smaller facets that are not perfectly aligned (e.g. Vrba et al., 2009). Zimmerman et al. (2020) observed the optical linear polarisation of a small sample of geostationary satellites during regular (relatively long lasting) glints, quasi-simultaneous with low resolution spectroscopic observations. In their data the authors observe that several satellites show a polarimetric signature around the glint, as expected from specular reflection off relatively large surfaces (e.g. Vrba et al., 2009), generally showing an increase in linear polarisation (tens of percent). In some other objects Zimmerman et al. (2020) did not detect such behaviour. Our MOP-TOP data has much better time resolution and sensitivity, allowing us to detect very small polarisation changes ($\sim 0.2\%$) on short timescales. This enables us to search for similar effects in micro-glints. Reflection off smaller satellite parts can in principle generate lower amplitude small glints; a distinguishing signature is how the faint glints and micro-glints behave over multiple nights as a function of solar phase angle (Hall & Kervin, 2013) and as a function of wavelength. As mentioned above, polarimetry can be an independent diagnostic. One possibility for the origin of the micro-glints is reflection of small reflecting components of the spacecraft bus or the antennas. Large sections of the bus are covered in multi-layer insulation (MLI), which can reflect highly specularly (e.g. Peltoniemi et al., 2021; Rodriguez et al., 2007), and may therefore give strong polarisation signatures under favourable reflection angles. A simplified laboratory setup has indeed shown strong polarisation features using a square bus model with Kapton (a polyimide film frequently used in MLI) as an example MLI (Beamer et al., 2018), with strong polarisation spikes near reflection angles close to the phase angle of our observation (i.e. the broad peaks in the lightcurve, e.g. at $\sim 54^\circ$, are reminiscent of the features seen in the analysis by Beamer et al., 2018).

Given the above it is somewhat surprising to only see weak evidence for polarisation spikes associated with the micro-glints. In some satellites, the MLI layer is fairly taut and smooth, in others it is more “wrinkly”. In the latter case, many individual reflecting facets/sections of MLI may contribute to the received light of the bus. As the phase angle changes, small sections may glint briefly, not unlike a disco ball. For the polarisation we expect the largest source of reflected light (the solar panels) to dominate the observed polarisation parameters at relatively large values of the phase angle, when the solar panel polarisation is high (e.g. Kosaka et al., 2020). On top of this baseline, the short glints from the MLI facets will give short polarisation spikes (as well as flux spikes), whose polarisation degree and angle depend on the material properties. In this wrinkly MLI scenario, there may be many superposed (micro-)glints and reflections (as seems supported by the flux lightcurve in

Figure 2), both diffuse and specular, whose polarimetric components sum up - but as the polarisation angles differ, this sum may result in a less obvious polarimetric signature at a given time. If the facets of MLI giving rise to the observed polarisation are fairly small compared to the size of the reflecting side of the satellite (and the other sources of polarised light, e.g. the solar panels, are large), we may expect the polarimetric signatures of the micro-glints to be relatively low in amplitude. In addition we note that at even with our high time resolution of 2 seconds, we may suffer from a degree of smearing of the signal. However since most of the micro-glints have resolved rise and fall times in the flux lightcurves, which have somewhat lower cadence than the MOPTOP sampling, this seems likely to be a relatively small effect.

In the MOPTOP interval we presented here (just 800 sec of data), the number of isolated, well characterisable micro-glints is relatively small. Future short timescale observations covering a much larger number of micro-glints are important to increase our sensitivity through statistics, and allow meaningful correlation studies. Observations covering a large phase angle range will be important, not just for the benefit of the modelling of the micro-glints but also to quantitatively model the dominant underlying components, such as the solar panels. Thor-6 is sufficiently bright over an entire night (Figure 2) that FAST mode observations are suitable over the entire night, i.e. fast timescale polarimetry is possible also at high phase angles. In addition, the datapoints obtained in FAST mode can be adaptively binned to decrease polarimetric errors at the expense of time resolution, allowing high accuracy measurements for somewhat fainter objects as well. We are also somewhat helped by the fact that the linear polarisation increases at increasing phase angle (Kosaka et al. 2020). At the brighter end, objects brighter than $\sim 5 - 6$ mag may saturate using FAST mode.

5.3. Future Prospects

Thor-6 is unresolved in our MOPTOP images (as expected; Hart et al., 2015). Some satellites in low and medium earth orbits will be resolvable by MOPTOP on the LT (or a similar instrument and telescope combination; a 5m satellite at 500km altitude can span $\sim 2''$): the pixel scale of MOPTOP is $0.42''$, and good and stable seeing conditions are common at La Palma. For these objects, imaging polarimetry during glints would yield a particularly rich amount of information, as the glint features can be directly attributed to specific sections of the spacecraft, removing some free parameters in a quantitative (e.g. Mueller matrix chain) modelling. As these objects move rapidly over the sky, demands on tracking speeds are much higher than for geostationary objects, but within reach of many modern telescope mounts.

Another route of future progress is the use of simultaneous multi-wavelength polarimetry, as the polarisation signal from (specular) reflection by a given material is strongly wavelength dependent. While spectro-polarimetry would provide the most ideal dataset, this would require either a large telescope or the use of long exposure times to obtain data with sufficiently small statistical errors per wavelength bin. Combined with inevitable overheads of most existing instruments

(e.g. CCD readout, waveplate rotation) and the need to accurately place and retain a target in the spectrograph slit, this makes it challenging to spectro-polarimetrically study the time resolved properties of micro-glints. Broadband imaging polarimetry is an easier option. MOPTOP currently has a single arm (Shrestha et al., 2020), and multi-colour data can therefore only be taken consecutively, through filter changes using the filter wheel. Future MOPTOP upgrades envisage the use of more than one arm, with light split by dichroic elements (as is done by the DIPol-UF imaging polarimeter for example, Piirola et al., 2020), which would enable strictly simultaneous multi-colour imaging polarimetry and greatly increase the science yield in the field of satellite observations. This is of particular interest for the proposed New Robotic Telescope (NRT), a robotic successor to LT with a primary mirror size of ~ 4 m, for which time-domain polarimetry is a key priority, and whose light collecting power would enable high-speed polarimetry of fainter satellites.

A third interesting possibility is observing the same satellite with two (or more) widely separated (in longitude or latitude) telescopes with polarimeters at the same time, using the same wavelength (filter). Of particular interest are situations when the telescopes are separated by distances of order the sun spot size of the (micro-)glints on the ground (e.g. Vrba et al., 2009), which would add a powerful diagnostic to identify the exact reflecting component responsible for the glinting behaviour, and its orientation. At large latitude differences, the viewing angles onto the reflecting areas orthogonal to the east-west direction is different enough that the resulting integrated polarisation outside of glints will be noticeable different. Given that the viewing angle offsets can be precisely calculated, this will provide a powerful additional constraint on polarisation model inversion fits. A first attempt to do this combining the LT (with MOPTOP) with the University of Leicester 0.5m telescope (with the LE2Pol optical dual-beam imaging polarimeter; Wiersema et al. in prep) was unsuccessful because of local COVID-19 access restrictions.

We finally point out that Thor-6 has been in space for a relatively long time (Section 2). The effects of the solar wind and intense ultraviolet radiation environment on the reflecting components of satellites is not well understood. The MLI and solar panels of the satellite may have aged considerably in this environment, with significant changes in their reflection properties compared to laboratory measurements. Future polarimetric observations of Thor-6, and polarimetric observations (at the same phase angles) of other Thor satellites with different times in orbit may help to diagnose the effects of aging.

6. Conclusions

In this paper we present a single 800 second observation of optical imaging polarimetry in the R band of geostationary satellite Thor-6, obtained with the MOPTOP instrument mounted on the 2m robotic Liverpool Telescope at La Palma (Spain). Our data probe short timescales (down to 2 seconds) at high polarimetric accuracy. We add to that dataset a high cadence optical lightcurve from the University of Warwick's test

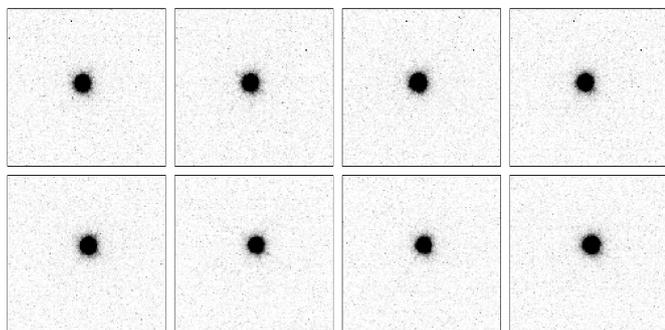


Fig. 6. Shown are the eight images belonging to the datapoint with unexpectedly low P_{lin} at phase angle 53.72 degrees (see Figure 4), as an example of a outlier datapoint. Shown are 120×120 pixel cut-outs (50.4×50.4 arcsec), with on the top row the four *cam2* images and on the bottom row *cam1*. There is no clear signature of a background star passing over the object.

telescope at La Palma, obtained simultaneously to the polarimetry. The lightcurve shows a large number of short timescale, high amplitude glints, often overlapping, which we refer to as micro-glints in this paper. This combined dataset is one of the most sensitive and highest cadence polarimetric observations of a geostationary satellite to date; the polarimetric observations overlap with a period of intense micro-glinting. In our MOPTOP data, the observed linear polarisation as a function of solar phase angle is dominated by a gradual evolution, which we may ascribe to reflection off the large solar panels of Thor-6. We can exclude strong polarisation features associated with micro-glints covered by our observation, but some faint features (bumps with a polarisation amplitude of a few tenths of percent) may be present in the lightcurves P_{lin} and polarisation angle. In particular, some increased scatter in the polarisation data is visible at the times of some of the micro-glint peaks. To establish correlation requires a future larger sample of micro-glints observed using high cadence polarimetry, and a greatly increased phase angle coverage. Our observation shows that the robotic LT with the MOPTOP instrument is a highly suitable combination to do this.

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