

A complete dataset of equatorial projections of Saturn's energetic neutral atom emissions observed by Cassini-INCA

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Key Points:

- All Ion and Neutral Camera observations of energetic neutral atoms from the Cassini Saturn tour are projected into Saturn's equatorial plane.
- The full set of projections in the form of daily fits files is accessible online for further scientific analyses.
- Different types of data contamination are identified and projections are flagged accordingly when loaded with the Python routine provided.

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Abstract

Observations of energetic neutral atoms (ENAs) are a useful tool for analyzing ion and neutral abundances in planetary magnetospheres. They are created when hot plasma, originating for example from magnetic reconnection sites, charge-exchanges with the ambient neutral population surrounding the planet. The motion of ENAs is not governed by the magnetic field, allowing remote imaging. During the Cassini mission, the Ion Neutral Camera (INCA) of the Magnetosphere Imaging Instrument (MIMI) collected vast amounts of hydrogen and oxygen ENA observations of Saturn's magnetosphere from a variety of different viewing geometries. In order to enable investigations of the morphology and dynamics of Saturn's ring current, it is useful to re-bin and re-project the camera-like views from the spacecraft-based perspective into a common reference frame. We developed an algorithm projecting INCA's ENA observations into a regular grid in Saturn's equatorial plane. With most neutrals and ions being confined into an equatorial rotating disc, this projection is quite accurate in both spatial location and preservation of ENA intensity, provided the spacecraft is located at large enough elevations. Such projections were performed for all INCA ENA data from the Cassini Saturn tour; the data is available for download together with a Python routine flagging contaminated data and returning detailed spacecraft geometry information. The resulting dataset is a good foundation for investigating for example the statistical properties of Saturn's ring current and its complicated dynamics in relation to other remote and in situ observations of, for example, auroral emissions and magnetotail reconnection events.

Plain Language Summary

Energetic neutral atom (ENA) observations contain useful information about ion and neutral abundances in planetary magnetospheres. They are created when energetic ions transfer their charge onto particles in a planet's neutral exosphere, leaving an energetic neutral with the original ion's energy and direction of motion. ENAs can be observed remotely, as their motion is no longer determined by the planetary magnetic field. Cassini's Ion and Neutral Camera (INCA) obtained such remote observations of Saturn's magnetosphere for more than a decade. To simplify large-scale analyses of this vast dataset, it is useful to re-bin and re-project all observations into a common reference frame, as the perspective of the imagery varies significantly depending on Cassini's position and attitude. Most ions and neutrals are confined to Saturn's equatorial plane due to the planet's rapid rotation, making this the source region of most ENAs and an ideal common reference frame. We developed an algorithm projecting all of INCA's ENA observations into a regular grid in Saturn's equatorial plane, creating a clean and easy-to-use dataset which can be utilized to investigate the morphology and dynamics of Saturn's ring current and its relation to, for example, ultraviolet auroral emissions and magnetic reconnection in the nightside magnetosphere.

1 Introduction

Energetic neutral atom (ENA) imaging is a useful technique for analyzing the interaction between hot space plasmas and the neutral gas in planetary magnetospheres. Energetic ions can undergo charge exchange with a neutral population, e.g., the planetary exosphere, to create an ENA particle which carries information on the original ion's energy and direction of motion. This ENA continues on its trajectory, unaffected by electromagnetic fields, and can be observed by a remote detector. Remote sensing of an ENA population hence provides spectral, compositional and pitch angle information of the source plasma. As the observed ENA flux is a line of sight integral dependent on both neutral and ion densities and their charge-exchange cross section, dynamics and gradients in ENA intensity may reflect not only dynamics of the ion source population but also the underlying neutral gas population (see, e.g., Brandt et al., 2018, and references therein).

While there were several instruments observing the terrestrial system, the Cassini orbiter carried the so far only ENA imager reaching the outer planets of our solar system. A flyby at Jupiter revealed ENA emissions from both the planet's upper atmosphere

69 and from a torus just outside the orbit of the icy moon Europa (e.g., Mauk et al., 2003;
70 Mitchell et al., 2004), details of which will be revealed by observations from the upcoming
71 JUPiter ICy moons Explorer (JUICE, Grasset et al. (2013)). Just recently, (Mauk et al.,
72 2020) demonstrated that the Juno spacecraft’s Jupiter Energetic particle Detector Instru-
73 ment is also capable of observing reasonable ENA foreground if the population of energetic
74 charged particles entering the detectors does not obscure the signal (e.g., as Juno moves
75 through high southern polar latitudes). This allowed for distinct observations of ENA
76 emission from the Io orbit, Europa orbit and Jupiter itself.

77 Upon reaching the Saturn system, years of ENA observations revealed a highly
78 dynamic magnetosphere driven both internally through Saturn’s rapid rotation and ex-
79 ternally through interaction with the solar wind. First observations revealed injections of
80 energetic plasma from the nightside, thought to be related to magnetotail reconnection
81 events and similar to substorms in the terrestrial magnetosphere (Mitchell et al., 2005).
82 The injected plasma is typically observed to corotate with Saturn for up to a full plane-
83 tary rotation or longer (e.g., Mitchell, Krimigis, et al., 2009; Carbary & Mitchell, 2014),
84 continuing to produce ENA emissions and forming the ring current.

85 Several studies have discovered periodicities in the ENA intensity indicative of ro-
86 tational modulation near the planetary rotation period (Krimigis et al., 2005; Paranicas
87 et al., 2005; Carbary et al., 2008, 2011; Carbary & Mitchell, 2017). These may be caused
88 by the periodic thinning and thickening of Saturn’s equatorial plasma sheet by planetary
89 period oscillation systems, two rotating systems for field-aligned currents which originate
90 near Saturn’s poles and span its entire magnetosphere (e.g., Jackman et al., 2016; Cowley
91 & Provan, 2017; Provan et al., 2018). This thickness modulation is also observed in the
92 statistical ENA intensity (Kinrade, Bader, et al., 2020) and may trigger periodic recon-
93 nection in the magnetotail (e.g., Bradley et al., 2018, 2020; Bader et al., 2019).

94 ENA emissions can often be mapped directly to ultraviolet auroral emissions through
95 field-aligned current systems connecting Saturn’s polar auroral ionospheres to the equato-
96 rial ring current plasma (Mitchell, Krimigis, et al., 2009; Kinrade, Badman, et al., 2020).
97 While observations of ion and electron beams on auroral field lines have been studied oc-
98 casionally (Mitchell, Kurth, et al., 2009; Bader et al., 2020), holistic investigations of the
99 relation between ENAs and the aurorae are needed to gain better understanding of the
100 physical process linking the two.

101 While many studies analyzed Cassini INCA ENA observations to reveal the intrigu-
102 ing and complex dynamics of Saturn’s magnetosphere, much of the INCA data obtained
103 during Cassini’s 13 years in orbit around Saturn is left nearly unexplored. This is partly
104 due to the highly variable viewing geometry and data quality of the ENA observations,
105 complicating especially the implementation of statistical studies. Our work is intended to
106 remove some of these barriers by projecting all INCA data into Saturn’s equatorial plane,
107 which is where the majority of ions and neutrals are concentrated and where most ENAs
108 are produced. This normalizes the dataset to a common reference frame and takes care of
109 proper data calibration and cleaning, making it easy to use and compare to other datasets.

110 In the next section a broad description of the INCA instrument is given, followed
111 by a detailed breakdown of the data calibration and projection procedures performed on
112 its observations. Section 4 provides an overview over the data coverage throughout the
113 Cassini mission, while section 5 informs about the different types of contamination ex-
114 pected in the dataset. In section 6 we investigate the response of the INCA instrument
115 to typical ENA fluxes, followed by a consideration of viewing geometry effects on the
116 observed ENA intensity in section 7. Lastly, we summarize our findings in section 8.

117 2 The MIMI-INCA instrument

118 The MIMI-INCA detector (Krimigis et al., 2004) was a time of flight (ToF) sensor capable
119 of measuring both ions and ENAs with energies from 7 keV to 3 MeV per nucleon, provid-
120 ing information on both the mass and the direction of motion and energy of each detected
121 particle. INCA had an instantaneous field of view (FOV) of $120^\circ \times 90^\circ$ and an angular
122 resolution of up to 64×64 pixels, i.e. $(120/64)^\circ \times (90/64)^\circ$. When operated in ion mode,
123 INCA was hence able to resolve a significant part of the local ion pitch angle distribution,

124 revealing for example field-aligned ion beams and conics on auroral field lines (Mitchell,
125 Kurth, et al., 2009; Bader et al., 2020).

126 Collimator plates mounted at the detector entrance could be charged to a large
127 potential, deflecting incoming charged particles and preventing them from entering the
128 detector. With this potential applied, only ENAs could be observed – providing remote
129 imagery of the global distribution of energetic ions using a technique known as ENA imag-
130 ing (e.g., Roelof, 1987).

131 Incoming particles triggered the time of flight sensor by penetrating a thin start
132 foil, producing secondary electrons which entered the start multichannel plate (MCP)
133 to provide the start time. The original incident particle then traveled further through
134 the instrument toward the end foil; upon impact secondary electrons were produced and
135 recorded via the stop MCP. The particle’s velocity could be obtained using its travel time
136 through the instrument and the distance between the start and stop foils. The mass of an
137 incident particle was determined by the pulse height of the MCP signal, as the number
138 of secondary electrons produced increases significantly with the particle mass. The pulse
139 height was sufficient to distinguish between the two most common neutrals in Saturn’s
140 magnetosphere, hydrogen and oxygen, but further mass discrimination was not feasible.
141 The particle energy could be obtained using the mass and velocity determined from the
142 measurements, and the direction of motion could be inferred from the particle’s impact
143 location on the MCP.

144 In order to produce sensible observations even when the spacecraft’s attitude was
145 not stable, INCA was equipped with two internal motion compensation algorithms. These
146 registered counts observed over the time of an exposure in an inertial coordinate image
147 buffer using quaternions transmitted to INCA from the spacecraft attitude system to take
148 into account any spacecraft motions.

149 When the spacecraft was rolling about its Z-axis (which it did for several hours dur-
150 ing many of the data downlink intervals), the flight software (FSW) knew where INCA’s
151 FOV was moving and so could keep a continuous de-spun buffer of “sky” pixels into which
152 counts were accumulated. Every 360° spin was divided into four quadrants, each of which
153 was transferred in telemetry frames to the spacecraft for downlink after INCA had fully
154 populated it. Since for a given sky pixel the amount of time spent exposed to each instru-
155 ment pixel was well known, the application of the flat-fielding matrix was well determined
156 such that all sky pixels could be accurately weighted by the instrument response to obtain
157 ENA intensities.

158 In the other instance the FSW did not know where the spacecraft would be moving;
159 so while counts could still be accumulated in inertial “sky” pixels, the weighting applied to
160 them for conversion to intensity was only approximated. In general, a sky pixel area larger
161 than the INCA FOV would have counts accumulated in it, and the FSW would choose the
162 centroid of that distribution that presumably contained the highest exposure times and
163 transmit this data into a telemetry frame. The time spent on each pixel was also retriev-
164 able from the quaternions. At the end of the exposure period, an image corresponding to
165 the size of the INCA FOV was extracted, centered on the average INCA boresight posi-
166 tion. This approach was useful when the spacecraft was either not articulating (in which
167 case the instrument coordinate system and the inertial sky system were identical), or when
168 the optical remote sensing instruments were producing mosaic images of a moon, Saturn,
169 or the rings, which typically meant rastering with rotations back and forth about the
170 spacecraft Z-axis interspersed with steps in angle about the spacecraft X-axis. Typically
171 these motions were on the order of or less than a single INCA pixel in extent, and so the
172 corrections between instrument and inertial systems were minimal. In cases where they
173 were large (for example during slews between different objects) the INCA images were not
174 generally very useful.

175 3 Data calibration and projection

176 The basis of our newly created dataset is the uncalibrated MIMI-INCA Planetary Data
177 System (PDS) archive C0-E/J/S/SW-MIMI-2-INCA-UNCALIB-V1.1. We first select all data of
178 interest by filtering for

Figure 1. Series of Cassini MIMI-INCA observations of hydrogen ENAs between 24-55 keV, averaged over 1 hour of exposure time. (a-d) ENA fluxes in the original detector grid. (e-h) Re-centered onto Saturn and rotated such that the north pole is up. The vertical line pointing upwards from the planet model indicates Saturn’s rotation axis (Z_{KSMAG}), the other line indicates the X_{KSMAG} axis directed toward the Sun. (i-l) Projections of these observations into Saturn’s equatorial plane as seen from above the north pole, with the Sun to the right. During the observation window shown here, Cassini moved from (X, Y, Z) (7.7, 1.7, 12.7) R_S to (5.1, 2.9, 12.6) R_S , essentially traversing high above the dayside magnetodisc toward Saturn’s dusk side.

- High-resolution type: **Spatial**
- High voltage (HV) state: 1 (deflection voltage turned on, includes ENA and excludes ion observations)
- Particle species: H for hydrogen-dominated, CNO for oxygen-dominated images
- Time of flight channel: 7 for high ToF (low energy range), 0 for low ToF (high energy range)

This selection returns four types of ENA imagery:

- Hydrogen, ToF channel 7 (24-55 keV), 32×32 pixels
- Hydrogen, ToF channel 0 (50-90 keV), 64×64 pixels
- Oxygen, ToF channel 7 (90-170 keV), 32×32 pixels
- Oxygen, ToF channel 0 (170-230 keV), 64×64 pixels

In theory all four image types are obtained during the same time, resulting in matching start and stop times for each exposure. However, we found that the HV state flags did sometimes not agree at the beginning or end of ENA imaging sequences, such that, e.g., hydrogen images were registered with HV state 1 (ENA images) while oxygen images exposed at the same time were registered with HV state 0 (ion images). These ambiguous cases were dropped from the dataset before any further processing was performed.

The raw counts of each image are calibrated and converted to differential energy flux using

$$\text{differential energy flux} = \frac{\text{counts} \times \text{flux factor matrix} \times \text{saturation ratio}}{\text{exposure time} \times \text{energy width}}, \quad (1)$$

where

- **differential energy flux** is in units of particles/cm²/s/sr/keV.
- **counts** is the raw number of counts from the PDS archive.
- **flux factor matrix** is the inverse of the pixel-wise product of the instrument geometric factor and efficiency, a matrix of the same size as the detector grid in units of /cm²/sr – different for each ENA image type and depending on observation mode (spinning vs. staring) as well as time of observation (efficiency degraded throughout the mission). These matrices are contained in the calibration files accompanying the raw data in the PDS archive.
- **saturation ratio** is the number of valid time of flight events divided by the number of events processed by the processing unit. This ratio is applied to reduce saturation effects.
- **exposure time** is the accumulation time of the image in seconds.
- **energy width** is the energy width of the ToF channel in keV.

These images are in the frame of the INCA detector; their perspective is hence dependent on spacecraft position and attitude. In order to remove this dependency, we project all observations into Saturn’s equatorial plane to obtain a consistent set of easily comparable observations. This projection is performed separately for each INCA exposure to minimize inaccuracies arising from spacecraft motion and attitude changes.

Figure 2. Sketch of two projected INCA pixels (colored boxes) intersecting the regular projection grid in the equatorial plane (black lines). Projection pixel A is located entirely within the purple projected INCA pixel polygon; it will simply be assigned the flux recorded by the purple INCA pixel. Projection pixel B on the other hand is intersected by the projections of both the purple and the orange INCA pixel; it will instead be assigned a flux value calculated from the purple and orange INCA pixels' recorded fluxes (mean value, weighted by the intersection area between the purple/orange INCA pixel and the projection pixel). The same applies for projection pixel C, where the final flux value is weighted by the relative size of the INCA pixels' intersection area and the white section (not intersected by an INCA pixel) is disregarded in the weighting. Lastly, projection pixel D will not be assigned a flux value as it is not intersected by any INCA pixel.

Figure 3. Timeline of Cassini's radial distance from Saturn and latitude above the equatorial plane through the entire Saturn tour. Blue shaded regions indicate days where MIMI-INCA ENA observations were obtained while Cassini was located between 4 and 100 Saturn radii from the equatorial plane.

215 We first determine Cassini's position and the INCA pointing at the central time
 216 of each exposure using the NAIF SPICE toolkit. Care needs to be taken to account for
 217 INCA's internal motion compensation, which may add offsets of an integer number of
 218 pixels in both the long and short dimensions of the INCA FOV. Once the INCA pointing
 219 has been adjusted accordingly, the geometric projection is done by tracing the line of sight
 220 of each INCA pixel into Saturn's equatorial plane. In order to conserve the shape of each
 221 INCA pixel in the projection, we trace its four corner view vectors as well as several sup-
 222 port vectors along each pixel edge. The intersection points of all these view vectors in the
 223 equatorial plane are then connected to form a polygon whose intersection with the base
 224 projection grid is determined.

225 Projection grid pixels entirely within the projected INCA pixel polygon are assigned
 226 the flux value of this INCA pixel, while projection grid pixels (partly) intersected by sev-
 227 eral INCA pixels are assigned the intersection area-weighted average of the intersecting
 228 INCA pixels' flux values; see Figure 2 for a graphical explanation. The projection grid is
 229 limited to $\pm 30 R_S$ distance from the center of Saturn in both the X_{KSMAG} and Y_{KSMAG}
 230 axes. The resulting projections are then saved for each exposure separately, combined into
 231 one `.fits` file per day and per ENA observation type, together with some metadata de-
 232 tailing the observation time, Cassini's location and some other parameters. These files are
 233 available for further analyses at [http://www.research.lancs.ac.uk/portal/en/datasets/
 234 cassini-inca-equatorial-ena-projections\(e9cd8998-75ab-4fff-8e6a-9bebb74ab54b\)
 235 .html](http://www.research.lancs.ac.uk/portal/en/datasets/cassini-inca-equatorial-ena-projections(e9cd8998-75ab-4fff-8e6a-9bebb74ab54b).html) (will be replaced with a DOI link once known). An example of raw and projected
 236 observations can be seen in Figure 1, where the top row shows some observations in the
 237 original INCA pixel grid and the bottom row shows the final equatorial projections ob-
 238 tained from them.

239 4 Available data

240 The INCA instrument performed ENA observations throughout most of Cassini's Saturn
 241 tour, from first approach to the final plunge into the upper atmosphere, without major
 242 interruptions. However, the accuracy and validity of the projections described in this study
 243 are dependent on the observation geometry of an exposure.

244 The two parameters of importance here are the spacecraft distance from and ele-
 245 vation above a projection pixel. The projected size of an INCA pixel increases with the
 246 square of the spacecraft distance and further depends on the cosine of the spacecraft el-

Figure 4. Example of an ion contamination event. (a-h) MIMI-INCA hydrogen neutral mode observations in two different energy ranges, some of which exhibit clear intensifications in pixels close to the local magnetic field direction (red cross), in the last image even a well resolved ion conic. (i) High energy resolution timeseries, the exposure times of panels (a-h) highlighted with gray shading near the top. The contamination events are characterized by increased fluxes in the higher energy bins (darker colors), times shaded red were identified as contaminated by our detection algorithm.

247 evation angle; lower elevation angles leading to increased stretching of pixels projected
 248 into the equatorial plane. Beyond certain spacecraft distances and/or elevation angles, the
 249 spatial resolution may hence be considered too low for reliable scientific analyses. On the
 250 other hand, data obtained too close to the equatorial plane may not produce representa-
 251 tive ENA intensity values due to the detector being located within the ENA source region.
 252 We hence recommend careful consideration of the viewing geometry of each exposure.

253 It is to note that many observations may have “good” and “bad” sections; e.g., with
 254 Cassini located at a reasonable distance above the dawn magnetodisc one could expect
 255 good resolution of the dawn, but stretched pixels with bad spatial resolution of the dusk
 256 ENA emissions. This means that instead of limiting the range of spacecraft distances
 257 and / or elevations relative to Saturn itself within which data is considered usable, it is
 258 preferable to limit the range of spacecraft distances and / or elevations relative to single
 259 projection pixels. The Python data loading function provided together with the equatorial
 260 projections is able to return the necessary geometric information on a per-pixel basis.

261 To give a rough idea of the availability of useful equatorial ENA projections, Figure 3
 262 shows Cassini’s orbital distance and elevation (latitude) relative to Saturn throughout
 263 its entire Saturn tour. All days during which ENA observations were obtained while
 264 Cassini was located at distances between 4 – 100 Saturn radii from the equatorial plane
 265 are shaded blue. While we notice lengthy gaps without any valid equatorial projections
 266 during Cassini’s equatorial orbits in 2006, 2010-2012 and 2015, there are plenty of observa-
 267 tions available from higher latitude orbits spread throughout the remaining mission. The
 268 dataset made available online includes projections of all mission data, regardless of viewing
 269 geometry.

270 5 Contamination handling

271 Virtually no dataset is free of bad data, and INCA data is no exception to that. Different
 272 kinds of contamination are known to occur in the INCA detector and should be removed
 273 from the dataset before performing scientific studies. A Python routine for loading the
 274 projected INCA data is provided with the dataset; it checks for all known kinds of con-
 275 tamination and flags possible events. We note that the contamination flagging is rather
 276 sensitive to make sure all contamination events are covered, so some data flagged as com-
 277 promised may be usable – whether this is the case for a certain observation is left to the
 278 user to decide.

279 The first type of bad data are “out of calibration” events. These occur when the
 280 INCA start and stop voltages are not in the proper calibration range, which is often re-
 281 lated to maintenance procedures but can also occur unplanned. A list of these events is
 282 provided by the instrument team together with INCA calibration files.

283 Data downlinked to Earth from the Cassini spacecraft can sometimes contain bit
 284 errors. These are usually identifiable as clear outliers in ENA intensity and are hence quite
 285 straightforward to filter out. Sunlight contamination similarly leads to enhanced intensities
 286 in affected detector regions and can be identified in the same way as bit errors; however,
 287 most or all of these events have already been removed from the dataset by the instrument
 288 team.

Figure 5. (a) Modeled INCA histograms for different constant incident fluxes. (b) INCA pixel sensitivity for a flux of 1 particle/cm²/s/sr/keV. (c-f) Histograms after rebinning of modeled INCA data, for hydrogen and oxygen in their respective energy ranges.

289 Another type of contamination is the occurrence of ion beam signatures in ENA
 290 data. While a strong potential could be applied to INCA’s collimator plates to prevent
 291 charged particles from entering the detector during ENA imaging phases, ions with en-
 292 ergies > 500 keV were still able to pass by design (Krimigis et al., 2004). However, even
 293 in the beginning of the mission ion beams could sometimes be observed to create signa-
 294 tures in ENA imagery when Cassini crossed auroral field lines. This indicated that either
 295 higher energy ions were present or the shielding was less efficient than assumed, or both.
 296 Moreover, a short developed in the high voltage circuit in 2015 such that the deflection
 297 potential could only reach half its intended strength, further reducing the performance of
 298 the ion deflector system. This means that especially toward the end of the Cassini mission
 299 many ENA observations contain features caused by ion beams.

300 An example of such contamination is shown in Figure 4. Shown are both INCA
 301 ENA images with high spatial resolution (panels 4a-h) and time series of high ToF res-
 302 olution INCA observations (panel 4i). Ion contamination events are visible as strong
 303 intensifications near the magnetic field-aligned direction (red cross). In panels 4b/f we can
 304 clearly identify a field-aligned ion beam whereas panels 4d/h appear to show an ion conic.
 305 The high ToF resolution data in panel 4i shows the measured “ENA” intensity in different
 306 energy ranges, and it is clear that ion signatures in ENA imagery correspond to sharp flux
 307 increases by sometimes more than an order of magnitude especially in the higher energy
 308 bins. The origin of these ion signatures is discussed in, e.g., Mitchell, Kurth, et al. (2009),
 309 Badman et al. (2012) and Bader et al. (2020).

310 This property is used to identify ion contamination events throughout the entire
 311 Cassini mission. This is done by comparing the 10-minute mean intensity with the 12-hour
 312 median intensity recorded in the high ToF channels; all times where the 10-minute mean
 313 is larger than twice the 12-hour median in one of the 90-149 keV or 149-227 keV energy
 314 channels and the relative increase in this channel is larger than in the 5-13 keV channel are
 315 flagged as likely ion contamination events.

316 A last type of bad data is the frequent occurrence of enhanced fluxes at the start
 317 and end of an ENA imaging sequence. These enhancements are likely caused by observa-
 318 tions being classified as “HV state 1” even though the deflection voltage was in the process
 319 of ramping up or down after or before ion observations were performed, leading to ions
 320 entering the detector and increasing the recorded fluxes. These events are identified in a
 321 similar fashion as ion contamination events described in the paragraphs above, bar the
 322 constraint of higher energy channels needing to show a larger flux increase than the lowest
 323 energy channel.

324 6 Instrument response and effects of the projection on 325 the flux histogram

326 Before performing scientific analyses on these data it is important to understand the
 327 caveats of INCA observations and the effects of performing projections as described in
 328 this study. Typical INCA exposures take some minutes to obtain, and during that time
 329 a single pixel usually collects at most a handful of hydrogen 24-55 keV counts. At higher
 330 energies the number of counts detected is certainly even lower, such that special care has
 331 to be taken when producing statistical averages such as shown in (Kinrade, Bader, et al.,
 332 2020).

333 To illustrate this, we reverse-engineered the number of counts expected within
 334 a typical 6 minute exposure from the instrument calibration procedure for a given
 335 ENA flux (see Figure 5b, showing the expected counts for a flux of 1 hydrogen

Figure 6. Mean ENA intensity observed between $7 - 11 R_S$ distance from Saturn’s center in the equatorial plane and at spacecraft elevation angles $> 50^\circ$. (a) Binned by spacecraft line of sight distance. (b) At spacecraft line of sight distances between $5 - 30 R_S$, observed at different “flow angles”. Flow angle is defined as the angle between the corotational particle flow in the ENA source region in the equatorial plane and the viewing direction of INCA. 0° flow angle hence corresponds to flow pointed away from INCA viewing direction and 180° to flow directly into the INCA detector.

particle/cm²/s/sr/keV in the 24-55 keV range). From these theoretical (non-integer) counts, a set of 10,000 observations with integer counts was created - i.e., for a pixel which should observe 0.76 counts in one 6 minute window, 7600 observations with one count and 2400 observations with 0 counts. Calibrating these theoretical observations with the typical routine results in certain histogram shapes which are shown in panel 5a.

We notice that in this case, for H 24-55 keV, high fluxes of 1 or 10 particles/cm²/s/sr/keV still result in a clear histogram peak at these ENA flux values, respectively. However, all lower fluxes have the same histogram shape which is entirely determined by the relative sensitivities of the detector pixels; the lowest measurable flux corresponds to one count in the most sensitive INCA pixel and is noticeable as a clear vertical boundary below which all histogram bins are empty (apart from the zero flux bin which is not shown here). At these fluxes, the real flux value is still determined to high accuracy even though there are no real histogram counts associated with it; the average strongly depends on the ratio of nonzero to zero detector count events. Statistical measures like the median or percentiles are hence meaningless and only the mathematical mean is a useful statistic.

The real data histograms however show observations of much lower fluxes, many orders of magnitude below the lowest fluxes observable by the detector (see also Kinrade, Bader, et al. (2020)). This is an effect of the projection procedure - one pixel in the projection grid may be intersected by more than one INCA pixel. The flux value assigned to this projection pixel then depends on the flux values of the intersecting INCA pixels and the relative area which is intersected. A zero-flux INCA pixel and a neighboring nonzero-flux INCA pixel intersecting the same projection pixel hence average to a flux value somewhere in between these two, depending on the ratio of intersected areas, which is eventually assigned to the intersected projection pixel. The effect of this was modeled by rebinning the theoretical data whose histogram is shown in panel 5a onto a smaller, randomly shifted grid. We observed that this creates an exponential tail toward low fluxes in the histograms, visible as a straight line in the log-log scaled histograms in panels 5c-f and in good agreement with the real data shown in Kinrade, Bader, et al. (2020).

7 Observation geometry effects: distance scaling and Compton-Getting effect

Lastly we investigate how ENA intensities recorded by INCA are influenced by the viewing geometry at the time of the observations. We hereby focus on two parameters: the line of sight distance between the instrument and the observed pixel in the equatorial plane, and the angle between the corotational plasma flow in the source region and INCA’s viewing direction.

This is done by selecting contamination-free observations of the ring current at radial distances between $7 - 11 R_S$ obtained from spacecraft elevations $> 50^\circ$ (above the observed pixel itself, not above the center of Saturn). These are then binned either by line of sight distance of the spacecraft from the source region or by “flow angle”, the angle between the corotational particle flow in the ENA source region in the equatorial plane and the viewing direction of INCA. Figure 6 shows the average ENA intensity depending on both of these parameters.

379 We notice that the spacecraft distance does not strongly affect the average ENA
 380 intensity (see panel 6a). This may be surprising; one would rightly expect the ENA intensi-
 381 ty to decrease $\sim r^{-2}$. However, while the physical intensity of a given part of the source
 382 region decreases at this rate, the size of the source region observed within one INCA pixel
 383 increases with $\sim r^2$ such that these two changes approximately cancel. This is only true
 384 for as long as a single INCA pixel captures a sufficiently homogenous ENA source region,
 385 but breaks down at larger distances – eventually INCA cannot properly resolve the ring
 386 current and an increase in pixel FOV will not increase the size of the ENA source observed
 387 within this pixel anymore; the ENA emitter is then more akin to a point source and the
 388 observed intensity should decrease roughly as $\sim r^{-2}$ as was shown in, e.g., Paranicas et al.
 389 (2005) where the total image average intensity was investigated.

390 Much of the variability visible in panel 6a is certainly attributable to temporal
 391 changes in the ENA source intensity and orbital bias. As most of the selected observations
 392 from Cassini’s Saturn tour were taken between roughly 10 – 20 R_S instrument line of sight
 393 distance, the average values shown are relatively smooth in between these distances but
 394 quite variable outside where far less observations are available and the average is likely to
 395 be more biased toward certain observation geometry or magnetospheric conditions.

396 Focusing on spacecraft distances between 10 – 20 R_S , we find that the mean hydro-
 397 gen 24 – 55 keV intensity shows only a slight decrease with increasing spacecraft distance,
 398 which appears to steepen for higher energies and heavier particles. Only for oxygen in the
 399 170 – 230 keV energy band does this decrease become significant though, likely a product of
 400 INCA’s decreased spatial resolution in this observation mode combined with the sparsity
 401 of such high energy particle counts which leads to observation characteristics more akin to
 402 spatially and temporally separate point sources rather than actual distributions of particles
 403 observed at lower energies.

404 Panel 6b shows how the mean ENA intensity depends on the angle between the
 405 source plasma motion relative to the detector viewing direction. It is important to inves-
 406 tigate this dependence in order to roughly quantify the overall impact of the Compton-
 407 Getting (CG) effect on the INCA dataset. The CG is a consequence of $E \times B$ and gradient-
 408 curvature drifts influencing the motion of energetic charged particles as they gyrate around
 409 magnetic field lines (Gleeson & Axford, 1968). A small “toward” or “away” motion of the
 410 plasma relative to the spacecraft shifts the detected energy of an ENA only slightly but
 411 in a power law type spectrum this is enough to cause a systematic measurement offset,
 412 known as CG effect. Specifically for INCA, Paranicas et al. (2005) investigated a single
 413 distant ENA population and concluded that a $\sim r^{-2}$ intensity decrease was the most im-
 414 portant effect on the signal and that the CG effect was minor for the species, energies, and
 415 distances they were considering.

416 Here it seems like the CG effect does not have a significant impact for the most part,
 417 as can be seen in panel 6b. Neither of the two hydrogen energy bands shows an increase in
 418 intensity toward larger “flow angles” (i.e., when the corotating plasma moves toward the
 419 detector), the curves are rather flat with some variability at more extreme values where
 420 less data is available. We note that the flow angle values shown are limited to 50 – 130°
 421 as the original data is limited to spacecraft elevations $> 50^\circ$ above the equatorially cor-
 422 rotating plasma. Oxygen observations do indeed show the expected dependency, with ENA
 423 intensities slightly increasing with flow angle; the effect is however very small and deemed
 424 negligible, especially for the qualitative analyses which these observations are mostly
 425 used for in the community. We hence conclude that at the spacecraft elevations $> 50^\circ$
 426 considered in this study the CG effect does not need to be taken into account for the de-
 427 termination of ENA intensities. At lower elevations the effect will of course be stronger as
 428 is discussed in, e.g., Paranicas et al. (2005), but equatorial projections should not be used
 429 from these perspectives anyway.

430 8 Conclusions

431 In this study we presented a new dataset containing equatorial projections of Cassini-
 432 INCA’s ENA observations of Saturn’s magnetosphere which is made available online for
 433 use in further scientific studies. We described the processing algorithm applied to the
 434 original raw data, including calibration and projection procedures, and summarized the

435 resulting dataset. Different types of data contamination have been identified and con-
 436 tamination events have been labeled in the final data product, paving the way for easier
 437 analyses of these valuable observations. We further investigated the effects of the projec-
 438 tion procedure applied to the original data and highlight different aspects which should be
 439 kept in mind when working with the projections we provide. This includes discussions on
 440 the low INCA count rates, the scaling of ENA intensity with line of sight distance and the
 441 impact of the Compton-Getting effect.

442 A first use case of this set is presented in Kinrade, Bader, et al. (2020), where we
 443 investigate the statistical distribution of ENAs in Saturn’s equatorial plane. We hope that
 444 this processed dataset will improve the accessibility of Cassini-INCA ENA observations
 445 and stimulate further important research on the dynamics of Saturn’s magnetosphere.

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 450 projections generated in this study are accessible at [http://www.research.lancs.ac.uk/
 451 portal/en/datasets/cassini-inca-equatorial-ena-projections\(e9cd8998-75ab-4fff-
 452 8e6a-9bebb74ab54b\).html](http://www.research.lancs.ac.uk/portal/en/datasets/cassini-inca-equatorial-ena-projections(e9cd8998-75ab-4fff-8e6a-9bebb74ab54b).html) (will be replaced with a DOI link once known). AB was
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